

Detection via simultaneous trajectory estimation and long time integration

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Abstract—In this work, we consider the detection of manoeuvring small objects with radars. Such objects induce low signal to noise ratio (SNR) reflections in the received signal. We consider both co-located and separated transmitter/receiver pairs, i.e., mono-static and bi-static configurations, respectively, as well as multi-static settings involving both types. We propose a detection approach which is capable of coherently integrating these reflections within a coherent processing interval (CPI) in all these configurations and continuing integration for an arbitrarily long time across consecutive CPIs. We estimate the complex value of the reflection coefficients for integration while simultaneously estimating the object trajectory. Compounded with this is the estimation of the unknown time reference shift of the separated transmitters necessary for coherent processing. Detection is made by using the resulting integration value in a Neyman-Pearson test against a constant false alarm rate threshold. We demonstrate the efficacy of our approach in a simulation example with a very low SNR object which cannot be detected with conventional techniques.

Index Terms—radar detection, coherent integration, non-coherent integration, bi-static radar, multi-static radar, target tracking, synchronisation.

I. INTRODUCTION

THE detection of manoeuvring and small objects with radars is a challenging task [1] and is a highly desirable capability in surveillance applications [2]. Radars emit modulated pulses towards a surveillance region and collect reflected versions of the transmitted waveforms from objects in this area. Small objects induce low signal-to-noise ratio (SNR) signals at the radar receiver. The decision on object presence is made by testing the hypothesis that the received signal contains reflections against the noise only signal hypothesis after the front-end input is filtered with a system response matching the probing waveform, which is known as the matched filter (MF) [3].

In order to detect low SNR objects, many such pulse returns (i.e., multiple measurements) need to be considered as each reflection is at a level similar to the noise background. The sufficient statistics of multiple pulse returns are found by summing the associated reflection coefficients across them, which is referred to as pulse integration [3, Chp.8]. This process is applied on the sampled outputs of the MF stage. These samples correspond to, in effect, measurements corresponding

to resolution bins in an equally divided range space. In conventional processing, beam-forming and Doppler processing with these samples are used to further segment the bearing and Doppler space into resolution bins and find the corresponding measurements. Conventional methods for integration over time such as coherent and non-coherent integration integrate pulse returns in the same range-bearing and Doppler bins across time. When objects manoeuvre, however, these reflections follow a trajectory across these bins, and, these methods fail to collect evidence on object existence for a long time due to not taking into account this trajectory. On the other hand, longer integration time provides higher probability of detection for a given false alarm rate, in principle.

One possible solution to providing long time integration for manoeuvring objects is to design filters with long impulse responses that match multiple pulse returns along a selection of possible trajectories (see, e.g., [4] & [5]). The number of filters required in this approach easily becomes impractically excessive with increasing integration time. An alternative approach is to employ a dynamic programming perspective and use a regular probing pulse MF to integrate its outputs along a trajectory estimated simultaneously which corresponds, in a sense, to on-line adaptive synthesis of long time MFs.

Trajectory estimation using the outputs of a pulse MF is often referred to as track-before-detect (see, for example, [6], [7]). The sample that corresponds to the true object kinematic state (i.e., location and velocity) is a complex value that is a sum of the reflection coefficient and background noise [8]. Most track-before-detect algorithms, on the other hand, use the modulus of the MF within models which describe the statistics of the modulus of the MF output. These models are averaged and hence cannot fully exploit the information captured by the measurements. For example, it is well known that the detection performance of these methods can be improved by also taking into account the phase of the data samples [9], in addition to the modulus.

The best achievable detection performance is obtained by coherent processing [3], in which one needs to estimate the complex reflection coefficient from the complex values of the MF outputs, the latter of which are processed by the aforementioned algorithms. This corresponds to using a non-averaged model in which the reflection coefficient is a random variable that remains the same during what is known as a coherent processing interval (CPI), and, is generated randomly for consecutive CPIs [8]. This is challenging partly because estimation of this quantity with a reasonable accuracy requires more samples than one can collect at the pulse-width sampling rate in a coherent processing interval (CPI) [10]. For example,

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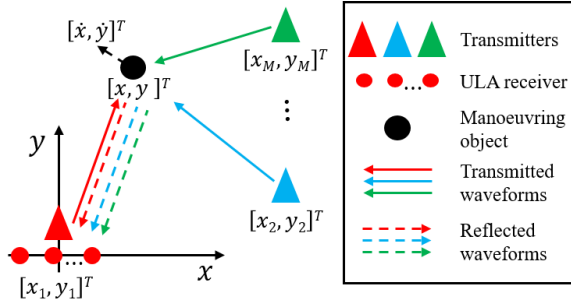


Fig. 1. Problem scenario: M transmitters and a ULA receiver to detect a small object located at $[x, y]^T$ with velocity $[\dot{x}, \dot{y}]^T$.

in [11], coherent processing and integration within a CPI is performed with a very high sampling rate that yields a large number of samples in the pulse interval.

In [12], we demonstrated that this can be remedied using a phased array receiver structure. In particular, we introduced a simultaneous trajectory estimation and long time integration algorithm in which the integrated value is then tested against a constant false alarm rate (CFAR) threshold for declaring the existence or otherwise of an object in a Neyman-Pearson sense. In [13], we extend this approach for separated transmitter/receiver pairs, i.e., bi-static channels, with an unknown time reference shift. We recover the synchronisation term by diverting simultaneous beams towards the tested point of detection and the remote transmitter thereby relaxing the commonly used assumption that the remote transmitters and the local receiver are synchronised (see, e.g., [14], [15]).

In this work, we provide a complete exposition of our long time integration and trajectory estimation approach in mono-static and bi-static configurations as well as the multi-static case. In particular, we consider the system structure in Fig. 1 where there are multiple transmitters using mutually orthogonal waveforms. The receiver is a ULA and has the full knowledge of the transmission characteristics except the time reference shift of the separately located transmitters. The front-end signals at the receive elements are the superposition of noise, signals from direct channels, and, reflections from objects.

We consider a long time likelihood ratio test conditioned on a trajectory in a kinematic state space, reflection coefficients, and, synchronisation terms as unknown parameters. In order to estimate the kinematic quantities, we use a Markov state-space model in which the object state consists of location and velocity variables. The measurement model of this state space model involves the radar ambiguity function parametrised on the aforementioned reflection coefficients. These coefficients are estimated by using an expectation-maximisation algorithm [16] realising a maximum likelihood (ML) approach within Bayesian filtering recursions for state trajectory estimate. We show that this is an empirical Bayesian method [17] for realising the update stage of the filter. When these ML estimates are reasonably accurate, the empirical Bayes update is an accurate approximation to the otherwise intractable filtering update equations. For synchronisation, we employ a digital beam-forming technique to simultaneously

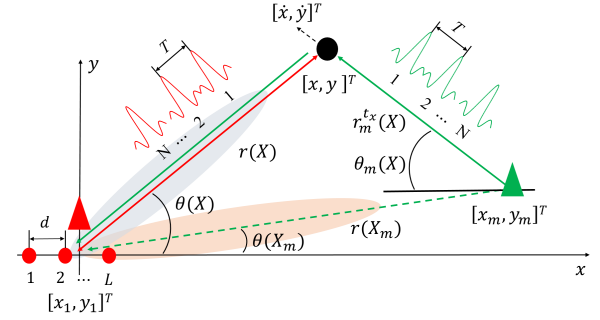


Fig. 2. Geometry of the problem: A ULA receiver co-located with a transmitter and another transmitter placed in a separate location on the 2D Cartesian plane. Both polar and Cartesian coordinate variables are depicted. Each transmitter emits N pulses in a CPI. The waveforms used are orthogonal.

divert beams towards both the test points of detection and the locations of the separately located transmitters in order to find the respective time reference shifts in the bi-static channels.

The resulting algorithm enables us to collect the entire evidence of object existence at the receiver by i) performing coherent integration in both mono-static and bi-static channels within a CPI, ii) non-coherently integrating across different (non-coherent) channels, e.g., local mono-static and remote bi-static channels, and, iii) continuing integration for an arbitrarily long interval that contains many CPIs. As a result, this approach enables us to detect manoeuvring and low SNR objects which cannot be detected using other techniques.

This article is organised as follows: Section II gives details of the problem scenario and introduces the mathematical statement of the problem. In Section III, we discuss trajectory estimation with the array measurements and detail the aforementioned empirical Bayes approach. In Section IV, we first introduce an expectation-maximisation (EM) algorithm for the ML estimation of the complex reflection coefficients. Then, we detail the ML estimation of the synchronisation term. We combine these estimators and specify the proposed detection scheme in Section V. The proposed detection algorithm is demonstrated in Section VI in comparison with a clairvoyant detector and a conventional scheme in a scenario with a manoeuvring and very low SNR object. Finally, we conclude in Section VII.

II. PROBLEM STATEMENT

Let us consider the problem scenario in Fig. 1 with a ULA receiver (depicted by red dots), and, M transmitters (depicted by triangles) one of which is co-located with the receiver forming a mono-static pair. The other transmitters are located elsewhere and form bi-static pairs with the receiver.

The receiver is comprised of L elements spaced with a distance of d which will be specified later in this section. Each element collects reflected versions of the transmitted waveforms emitted by both the co-located and the separately located transmitters thereby forming mono-static and bi-static pairs, respectively.

A. Spatio-temporal signal model

A detailed model for the signals induced at the receiver array by reflections from an object is as follows: We consider an

interval of time in which each transmitter emits N consecutive waveforms $\tilde{u}_m(t)$ separated by a time length of T after modulating with a common carrier that has an angular frequency of $\omega_c = 2\pi f_c$. The m th transmitted signal is therefore given by

$$u_m(t) = \text{Re} \left\{ \sum_{n=0}^{N-1} \tilde{u}_m(t - nT) e^{j\omega_c t} \right\}, \quad (1)$$

where $\text{Re}\{\cdot\}$ denotes the real part of its input complex argument and T is known as the pulse repetition interval (PRI).

We assume that $\{\tilde{u}_m\}_{m=1}^M$ is an orthogonal set of waveforms of pulse duration T_p and bandwidth B , i.e.,

$$\begin{aligned} \langle \tilde{u}_m(t), \tilde{u}_{m'}(t) \rangle &\triangleq \int_0^{T_p} \tilde{u}_m(t) \tilde{u}_{m'}^*(t) dt \\ &= \delta_{m,m'} \end{aligned} \quad (2)$$

for $m, m' \in \{1, \dots, M\}$, where $\delta_{m,m'}$ is Kronecker's delta function.

Use of such orthogonal transmit waveforms underlies the vision of multiple-input multiple-output (MIMO) radars [18], [19] a particular configuration of which is, hence, the system considered here. Design of orthogonal sets for MIMO sensing was investigated with various objectives such as maximisation of diversity [20] and waveform identifiability [21]. In this work, we consider a narrowband regime in which frequency division multiplexing can be used to achieve orthogonality in practice.

In order to specify the received signal at the array elements, let us consider the geometry of the problem which is illustrated in Fig. 2 for $M = 2$ transmitters. The receiver array measures the superposition of signals from different channels which are depicted by coloured lines. In particular, there are i) a local (mono-static) channel (red line), ii) a remote (bi-static) channel (green line), and, iii) a direct channel from the remote transmitter (green dashed line). The first two are reflection channels propagating the reflected waveforms from the object (black circle) towards the receiver array. These channels can be fully separated given the array data by exploiting the orthogonality of the waveforms over time and the capability of spatial filtering thereby diverting multiple beams towards arbitrary arrival angles, simultaneously. These points will become clear in the sequel.

Let us model the signals in the reflection channels. We assume that the reflectivity of the object remains coherent (i.e., unchanged) during the collection of reflections from the N pulses in (1). Such a time interval is known as a coherent processing interval (CPI). Modelling of the direct channel signals is introduced later in Section IV-B.

The kinematic state of the reflector (depicted by a black dot) in the 2D Cartesian plane is given by $X = [x, y, \dot{x}, \dot{y}]^T$, where $[x, y]^T$ is the location, $[\dot{x}, \dot{y}]^T$ is the velocity, and T denotes vector transpose. The distance of X to the receiver is related to pulse time of flights. The overall distance a pulse emitted by the m th transmitter at $[x_m, y_m]^T$ and reaches the receiver at $[x_1, y_1]^T$ after getting reflected at $[x, y]^T$ is given by

$$R_m(X) = R_m^{tx}(X) + R(X) \quad (3)$$

$$\begin{aligned} R_m^{tx}(X) &\triangleq \sqrt{(x - x_m)^2 + (y - y_m)^2}, \\ R(X) &\triangleq \sqrt{(x - x_1)^2 + (y - y_1)^2} \end{aligned}$$

where R_m and R denote the distance from the object to the m th transmitter and to the receiver, respectively.

The corresponding time of flights are found as

$$\begin{aligned} \tau_m(X) &= \tau_m^{tx}(X) + \tau(X) \\ \tau_m^{tx}(X) &\triangleq \frac{R_m^{tx}(X)}{c}, \quad \tau(X) \triangleq \frac{R(X)}{c}, \end{aligned} \quad (4)$$

where $c \approx 3.0 \times 10^8$ m/s is the speed of light.

The velocity of the object induces an angular frequency shift on reflections which is known as the Doppler shift. This quantity is given by

$$\begin{aligned} \Omega_m(X) &= \frac{2\pi T}{\lambda_c} \left(\dot{x} \times (\cos \theta(X) + \cos \theta_m(X)) \right. \\ &\quad \left. + \dot{y} \times (\sin \theta(X) + \sin \theta_m(X)) \right), \end{aligned} \quad (5)$$

where θ and θ_m are the angle of arrival (AoA) of the reflections to the receiver and the bearing angle of the object with respect to the m th transmitter, respectively. These quantities are given by

$$\begin{aligned} \theta(X) &= \arctan(y_1 - y)/(x_1 - x) \\ \theta_m(X) &= \arctan(y_m - y)/(x_m - x). \end{aligned} \quad (6)$$

For narrowband reflections, the signals induced at the array elements are characterised by a spatial steering vector as a function of θ which is given by [22, Chp.2]

$$\mathbf{s}_s(\theta) = \left[1, e^{-j\omega_c \frac{d}{c} \sin \theta}, \dots, e^{-j\omega_c (L-1) \frac{d}{c} \sin \theta} \right]^T,$$

where d is the separation between the array elements selected as half of the carrier wavelength, i.e., $d = \lambda_c/2$. Substituting this quantity together with $c = \lambda_c \times f_c$ in the equation above leads to

$$\mathbf{s}_s(\theta) = \left[1, e^{-j\pi \sin \theta}, \dots, e^{-j(L-1)\pi \sin \theta} \right]^T. \quad (7)$$

The superposition of the reflections after demodulation at the receiver is given using (7) and (5) by

$$\begin{aligned} \mathbf{z}(t) &= \mathbf{s}_s(\theta) \sum_{m=0}^{M-1} \sum_{n=0}^{N-1} \alpha_m e^{jn\Omega_m} e^{-j\omega_c(\tau_m + \Delta t_m)} \\ &\quad \times \tilde{u}_m(t - \tau_m - \Delta t_m - nT), \end{aligned} \quad (8)$$

where α_m is a complex coefficient modelling the reflectivity in the m th channel, and, τ_m is the time of flight of a pulse given in (4). Here, Δt_m is an unknown time shift modelling the time reference difference between the m th transmitter and the receiver (i.e., a synchronisation term).

The reflections in the received signal are optimally searched by matched filtering [8], i.e., by convolving the input with inverted versions of the probing waveforms. In our scenario, this corresponds to a bank of M filters, (see, e.g. [19, Chp.3]). Owing to the orthogonality (asserted by (2)), the M channels

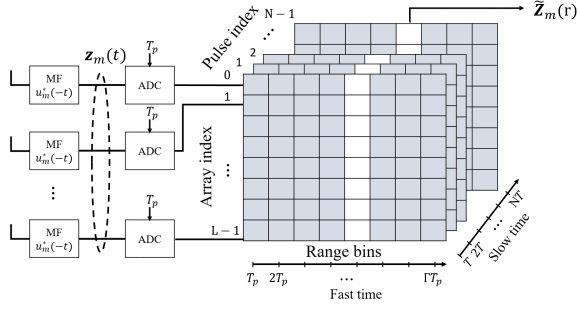


Fig. 3. Data acquisition in the m th channel: Sampled version of the received signal within a CPI as a radar data cube. The output of the matched filter is sampled and arranged in array index, fast time and slow time axis.

in (8) will have been separated at the filter outputs¹. The output of the m th filter is given by

$$\begin{aligned} \mathbf{z}_m(t) &\triangleq \mathbf{z}(t) * \tilde{u}_m(-t) \\ &= \alpha_m \mathbf{s}_s(\theta) \sum_{n=0}^{N-1} e^{jn\Omega_m} e^{-j\omega_c(\tau_m + \Delta t_m)} \\ &\quad \times \Lambda_m(t - \tau_m - \Delta t_m - nT). \end{aligned} \quad (9)$$

where $*$ denotes convolution and $\Lambda_m(\cdot)$ is the auto-correlation of the m th waveform given by

$$\begin{aligned} \Lambda_m(t) &= \tilde{u}_m(t) * \tilde{u}_m^*(-t) \\ &= \int_0^{T_p} \tilde{u}_m(t') \tilde{u}_m^*(t' - t) dt'. \end{aligned} \quad (10)$$

This output is sampled with a period that equals to the pulse duration T_p . Let us assume that T is an integer multiple of T_p , i.e., $T = \Gamma \times T_p$ where $\Gamma \in \mathbb{Z}^+$. $\Gamma \times N$ samples of this discrete time vector sequence is given by

$$\begin{aligned} \mathbf{z}_m[\gamma] &= \mathbf{z}_m(\gamma T_p), \quad \gamma = 1, \dots, \Gamma \times N, \\ &= \alpha_m s(\Delta t_m) \mathbf{s}_s(\theta) \mathbf{s}_t(\tau_m, \Omega_m)^T \\ &\quad \times \begin{bmatrix} \Lambda_m(\gamma T_p - \tau_m - \Delta t_m) \\ \Lambda_m(\gamma T_p - \tau_m - \Delta t_m - T) \\ \vdots \\ \Lambda_m(\gamma T_p - \tau_m - \Delta t_m - (N-1)T) \end{bmatrix}, \end{aligned} \quad (11)$$

where

$$s(\Delta t_m) \triangleq e^{-j\omega_c \Delta t_m}, \quad (12)$$

$$\mathbf{s}_t(\tau', \Omega') \triangleq e^{-j\omega_c \tau'} \times [1, e^{j\Omega'}, \dots, e^{j(N-1)\Omega'}]^T. \quad (13)$$

The term \mathbf{s}_t will be referred to as the temporal steering vector.

Next, this vector sequence is arranged as a cube by folding the two dimensional data array in (11) in lengths of Γ samples. The n th layer of the resulting cube corresponds to the samples collected between the n th and the following pulse, i.e.,

$$\mathbf{C}_m[n] \triangleq [\mathbf{z}_m[n\Gamma], \mathbf{z}_m[n\Gamma + 1], \dots, \mathbf{z}_m[(n+1)\Gamma - 1]]$$

¹Perfect orthogonality of waveforms might not be achievable in practice, nevertheless, design of waveforms with a fairly small mutual cross-correlation has been a productive research area which is also discussed, for example, in [19, Chp.2].

This processing chain is illustrated in Fig. 3 together with the cube $\mathbf{C}_m[n]$ which is also known as the radar data cube [3]. The axes of this cube are array index, slow time and fast time. In the fast time axis, we have Γ samples of the filter output, each of which is associated with a time delay of the reflected signal. These time delays correspond to time of flights which can easily be converted to range values using (4). As a result, N array measurements from range bin r is a slice along the slow time axis given by

$$\begin{aligned} \tilde{\mathbf{Z}}_m(r) &\triangleq [\mathbf{z}_m[r], \mathbf{z}_m[\Gamma + r], \dots, \mathbf{z}_m[(N-1)\Gamma + r]] \\ &= \alpha_m s(\Delta t_m) \mathbf{s}_s(\theta) \mathbf{s}_t(\tau_m, \Omega_m)^T \\ &\quad \times \Lambda_m(rT_p - \tau_m - \Delta t_m). \end{aligned} \quad (14)$$

For convenience regarding the notation in the rest of this article, we stack columns of $\tilde{\mathbf{Z}}_m(r)$ and form a $LN \times 1$ data vector. Before specifying this vector, let us combine the signal model in a single entity as a function of the reflector kinematic state X which induces the signals and the range bin r which is the measurement index:

$$\begin{aligned} \mathbf{s}_m(r, X) &\triangleq s(\Delta t_m) \mathbf{s}_s(\theta(X)) \otimes \mathbf{s}_t(\tau_m(X), \Omega_m(X)) \\ &\quad \times \Lambda_m(rT_p - \tau_m(X) - \Delta t_m) \end{aligned} \quad (15)$$

where \otimes denotes the Kronecker product operator. Here, $X = [x, y, \dot{x}, \dot{y}]$ is related to the data vector through the associated time of flight τ_m and (R, θ, Ω_m) found by evaluating (3)–(6). The r th column measurement vector for the hypotheses that a reflector object exist at X and the null hypothesis are hence given by

$$\mathbf{Z}_m(r) = \begin{cases} \alpha_m \mathbf{s}_m(r, X) + \mathbf{n}_m(r) & , H_1 \text{ holds,} \\ \mathbf{n}_m(r) & , H_0 \text{ holds,} \end{cases} \quad (16)$$

where $\mathbf{n}_m(r) \sim \mathcal{CN}(\cdot; \mathbf{0}, \Sigma_m)$ models the noise background of the m th channel and is a complex Gaussian random variable with zero mean and covariance of Σ_m .

Note that, for the (local) mono-static channel $m = 1$, and, (16) is found for $\tau_m = 2R/c$ and the synchronisation term $\Delta t_m = 0$ in (15). For $m > 1$, the measurement vectors are associated with the (remote) bi-static channels, and Δt_m is non-zero and unknown.

B. Problem definition

We would like to perform a hypothesis test based on the measurement model in (16). These measurements are complex numbers and we are interested in the evaluation of the sufficient statistics for the two hypothesis. Detection/processing using complex measurements are often referred to as coherent detection/processing and conventionally the input is the same resolution bin over multiple pulse returns [3]. Therefore, in order for this operation to maintain coherence, the target position should not be changed.

In order to extend coherent processing to the case of manoeuvring objects and remote transmitters, we introduce the mathematical statement of the problem as evaluation of a likelihood-ratio i) using complex versions of measurements (as

opposed to, for example, using only their moduli) for all M reflection channels, and, ii) for a time window of K CPIs given an object trajectory $\{X_k\}_{k=1}^K$ where $X_k = [x_k, y_k, \dot{x}_k, \dot{y}_k]^T$ is the object kinematic state at the k th CPI. This likelihood ratio will then be tested against a threshold in a Neyman-Pearson sense [23, Chp.3]. The detector we consider hence takes the form

$$L(\mathbf{Z}_{1:K}, \dots, \mathbf{Z}_{M,1:K} | X_{1:K}, \boldsymbol{\alpha}, \Delta \mathbf{t}) \underset{H_0}{\overset{H_1}{\geq}} \mathcal{T} \quad (17)$$

where $\mathbf{Z}_{m,1:K}$ are the data cubes for channel m over $k = 1, \dots, K$. Here, $\boldsymbol{\alpha}$ and $\Delta \mathbf{t}$ are reflectivity and synchronisation vectors across the channels, respectively, defined by

$$\boldsymbol{\alpha} \triangleq [\alpha_{1,1}, \dots, \alpha_{1,K}, \dots, \alpha_{M,1}, \dots, \alpha_{M,K}],$$

$$\Delta \mathbf{t} \triangleq [\Delta t_1, \Delta t_2, \dots, \Delta t_M].$$

In order to carry out the test in (17), the trajectory $X_{1:K}$ needs to be estimated. This is also referred to as tracking and is the subject of Section III along with estimation of the reflection coefficients $\boldsymbol{\alpha}$. Algorithmic strategies for estimating the synchronisation term $\Delta \mathbf{t}$ are introduced in Section IV. These results are combined in Section V and threshold selection is detailed in order to evaluate the detection test in (17).

C. Sufficient statistics for the likelihood ratio

The likelihood ratio on the left hand side of (17) factorises over as the noise samples for different CPIs are also independent. Each time term also factorises over channel likelihood ratios as the related parameters are independent, i.e.,

$$L = \prod_{k=1}^K \prod_{m=1}^M \frac{l(\mathbf{Z}_{m,k} | X_k, \alpha_{m,k}, \Delta t_m, H = H_1)}{l(\mathbf{Z}_{m,k} | X_k, \Delta t_m, H = H_0)}. \quad (18)$$

These measurements also satisfy a locality property in that the number of range bins which are associated with X_k is limited by the support of Λ in (10) which is of duration $2T_p$. Let us define the (range) extend of an object as

$$\mathcal{E}_m(X_k) = \begin{cases} \{r_{m,k}, r_{m,k} + 1\}, & r_{m,k}T_p < \tau_m(X_k) + \Delta t_m \\ \{r_{m,k}\}, & r_{m,k}T_p = \tau_m(X_k) + \Delta t_m \\ \{r_{m,k} - 1, r_{m,k}\}, & r_{m,k}T_p > \tau_m(X_k) + \Delta t_m \end{cases} \quad (19)$$

where

$$r_{m,k} \triangleq \left\lceil \frac{\tau_m(X_k) + \Delta t_m}{T_p} \right\rceil, \quad (20)$$

with $\lceil \cdot \rceil$ denoting the nearest integer function, and $\tau_m(X_k) + \Delta t_m$ gives the time of flight in the m th channel associated with the object state X_k . This range bin has the highest signal-to-noise ratio (in the m th channel) given that Λ as a time auto-correlation function typically vanishes towards tails.

As a result, the likelihood ratio in (18) further decomposes into factors over range bins as

$$L = \prod_{k=1}^K \prod_{m=1}^M \prod_{r \in \mathcal{E}_m(X_k)} \frac{l(\mathbf{Z}_{m,k}(r) | X_k, \alpha_{m,k}, \Delta t_m, H = H_1)}{l(\mathbf{Z}_{m,k}(r) | H = H_0)}, \quad (21)$$

The numerator terms in (21) can easily be found using the distribution of the noise in the signal model in (16) as

$$l(\mathbf{Z}_{m,k}(r) | X_k, \alpha_{m,k}, \Delta t_m, H = H_1) = \mathcal{CN}(\mathbf{Z}_{m,k}(r); \alpha_{m,k} \mathbf{s}_m(r, X_k), \Sigma_m). \quad (22)$$

The denominator in (21) regarding the noise only hypothesis is nothing but the noise density evaluated at $\mathbf{Z}_{m,k}(r)$. Therefore, the instantaneous likelihood ratio in (21) after substituting from (22) and the noise distribution is found as

$$\begin{aligned} & \frac{L(\mathbf{Z}_{m,k}(r) | X_k, \alpha_{m,k}, \Delta t_m)}{\mathcal{CN}(\mathbf{Z}_{m,k}(r); \alpha_{m,k} \mathbf{s}_m(r, X_k), \Sigma_m)} \\ & \triangleq \frac{\mathcal{CN}(\mathbf{Z}_{m,k}(r); \alpha_{m,k} \mathbf{s}_m(r, X_k), \Sigma_m)}{\mathcal{CN}(\mathbf{Z}_{m,k}(r); \mathbf{0}, \Sigma_m)} \\ & = \exp \left\{ 2 \operatorname{Re} \left\{ \alpha_{m,k}^* \mathbf{s}_m(r, X_k)^H \Sigma_m^{-1} \mathbf{Z}_{m,k}(r) \right\} \right\} \\ & \times \exp \left\{ -|\alpha_{m,k}|^2 \mathbf{s}_m(r, X_k)^H \Sigma_m^{-1} \mathbf{s}_m(r, X_k) \right\}, \quad (23) \end{aligned}$$

where $(\cdot)^H$ is the Hermitian of its argument, $\operatorname{Re}\{\cdot\}$ takes the real part of its complex argument, $(\cdot)^*$ denotes conjugate and $|\cdot|$ denotes modulus of a complex variable, respectively.

The likelihood ratio evaluation given in (23) is advantageous in that only a linear operation needs to be performed on the measurements which is in the form of a whitening transform with the inverse noise covariance followed by an inner product with the signal model. Because the signal model involves the spatial steering vector in (7), this inner product effectively performs beam-forming on the measurements filtering out contributions of other objects at the same range. Note that a second filtering is with respect to the Doppler as the temporal steering vector (13) is also in the signal model.

III. SIMULTANEOUS TRACKING AND REFLECTION COEFFICIENT ESTIMATION

In this section, we consider estimation of the object trajectory $X_{1:K}$ using coherent pulse returns during a CPI. Object trajectories are modelled as random vector sequences generated by a Markov state space model [24], i.e.,

$$X_{1:K} \sim p(X_1) \prod_{k=2}^K p(X_k | X_{k-1}), \quad (24)$$

where the Markov transition density is selected as

$$p(X_k | X_{k-1}) = \mathcal{N}(X_k; F X_{k-1}, Q)$$

$$F = \begin{bmatrix} 1 & 0 & \Delta & 0 \\ 0 & 1 & 0 & \Delta \\ 0 & 0 & 1 & 0 \\ 0 & 0 & 0 & 1 \end{bmatrix}, \quad (25)$$

where Δ is the time interval between two consecutive pulse train transmissions (or, the illumination period), F models constant velocity motion, and Q is the covariance matrix specifying the level of the process noise modelling unknown manoeuvres.

The initial distribution $p(X_1)$ is selected as a uniform distribution over the range-bearing interval for the detection test.

These intervals often correspond to radar specific resolution bins. Let us denote the corresponding bounded set in the state space by \mathcal{B} , and a uniform distribution on \mathcal{B} by $U_{\mathcal{B}}$. Therefore,

$$p(X_1) = U_{\mathcal{B}}(X_1). \quad (26)$$

Sequential estimation of $X_{1:K}$ as data cubes arrive is performed by using Bayesian recursive filtering [24]. Suppose we have the given distribution of the state variable at the time step $k-1$ based on all the measurements collected up to and including CPI $k-1$, i.e., $p(X_{k-1}|\mathbf{Z}_{1:k-1})$. In order to update this prior information with the measurement at the k th CPI, first, the Chapman-Kolmogorov equation is realised and a prediction density is found as

$$p(X_k|\mathbf{Z}_{1:k-1}) = \int p(X_k|X_{k-1})p(X_{k-1}|\mathbf{Z}_{1:k-1})dX_{k-1}, \quad (27)$$

where the first term inside the integral is the Markov transition given by (25).

The update stage of the filtering is given by multiplying this prediction and the measurement likelihood together with marginalising out all other variables, i.e.,

$$p(X_k|\mathbf{Z}_{1:k}) \propto \int_{\alpha_k} \int_{\Delta t} l(\mathbf{Z}_k|X_k, \alpha_k, \Delta t) \times p(\alpha_k)p(\Delta t)p(X_k|\mathbf{Z}_{1:k-1})d\alpha_k d\Delta t, \quad (28)$$

where $p(\alpha_k)$ and $p(\Delta t)$ are prior densities for the reflection coefficient and the synchronisation term, respectively.

The measurement likelihood in (28) is the product of the numerator terms in the likelihood ratio in (21) over the object's range bins and channels for the time step k , i.e.,

$$l(\mathbf{Z}_k|X_k, \alpha_k, \Delta t) \propto \prod_{m=1}^M \prod_{r \in \mathcal{E}_m(X_k)} l(\mathbf{Z}_{m,k}(r)|X_k, \alpha_{m,k}, \Delta t_m, H = H_1) \quad (29)$$

and is easily computed by evaluating complex Gaussian densities as discussed in Section II-C.

The marginalisation of the reflection coefficients and synchronisation terms, however, is not straightforward: First, one needs to select prior densities for these terms. One reasonable approach is to use a non-informative prior such as Jeffrey's prior [25, Chp.5]. These priors are useful when they lead to tractable computations in (28) (see, e.g., [26]). In our problem setting, however, Jeffrey's priors for the reflection coefficients and the synchronisation terms are constant, and, do not help in finding a tractable form for the full Bayesian update in (28).

In order to tackle this challenge, we use an empirical Bayes approach [17]. These methods approximate the integration in (28) by solving an optimisation problem for finding the likely values of the unknowns and evaluating the integrand at those values. In other words, (28) is rewritten as

$$p(X_k|\mathbf{Z}_{1:k}) = \int_{\alpha_k} \int_{\Delta t} p(X_k|\mathbf{Z}_{1:k}, \alpha_k, \Delta t)p(\alpha_k, \Delta t|\mathbf{Z}_{1:k})d\alpha_k d\Delta t. \quad (30)$$

Here, the reflection coefficients and the synchronisation terms act as model parameters to be selected and the second

term inside the integration is similar to a prior for them. Because this prior is conditioned on the measurements, more probability mass should be concentrating around the maximum likelihood (ML) estimates of these values. Let us select this density as

$$\begin{aligned} p(\alpha_k, \Delta t|\mathbf{Z}_{1:k}) &= p(\alpha_k|\mathbf{Z}_{1:k})p(\Delta t|\mathbf{Z}_{1:k}) \\ p(\alpha_k|\mathbf{Z}_{1:k}) &\leftarrow \delta_{\hat{\alpha}_k}(\alpha_k) \\ p(\Delta t|\mathbf{Z}_{1:k}) &\leftarrow \delta_{\hat{\Delta t}}(\Delta t), \end{aligned} \quad (31)$$

where \leftarrow denotes assignment and δ is Dirac's delta distribution. In other words, we select the model densities given the measurements as a Dirac's delta distribution concentrated around their ML estimates $\hat{\alpha}_k$ and $\hat{\Delta t}$, respectively.

After substituting from the empirical priors in (31) into (30), one obtains the empirical Bayes update as

$$p(X_k|\mathbf{Z}_{1:k}) \propto l(\mathbf{Z}_k|X_k, \hat{\alpha}_k, \hat{\Delta t})p(X_k|\mathbf{Z}_{1:k-1}) \quad (32)$$

where \propto denotes approximate proportionality. The approximation accuracy is better when these ML estimates are obtained using informative likelihoods (as quantified by their Fisher information) and equivalently have small CRLBs.

We will detail ML estimation of the reflection coefficients α and the synchronisation terms Δt in Section IV. For the remaining part of this section, let us assume that these estimates are given.

For realising the recursive filtering, a sequential Monte Carlo (SMC) approach known as the particle filter is used [27]. In particular, we use a bootstrap filtering approach for estimating the object trajectory.

The prediction stage at the time step $k=1$ is realised by forming a regular grid of P points over \mathcal{B} representing samples generated from the initial state distribution in (26). These points constitute an equally weighted set of particles. For $k > 1$, we will have found weighted samples, or, particles, representing the state posterior in the previous step. Let us denote this set by

$$\left\{ X_{k-1}^{(p)}, \zeta_{k-1}^{(p)} \right\}_{p=1}^P,$$

where $\zeta_{k-1}^{(p)}$ is the weight of the p th sample. The prediction stage is then realised by sampling from the Markov transition as

$$X_{k|k-1}^{(p)} \sim p(\cdot|X_{k-1}^{(p)}), \quad p = 1, \dots, P. \quad (33)$$

The weights of these samples in the particle set $\{X_{k|k-1}^{(p)}, \zeta_{k|k-1}^{(p)}\}$ is given by

$$\zeta_{k|k-1}^{(p)} = \zeta_{k-1}^{(p)}, \quad (34)$$

in order for this set to represent the prediction density in (27).

In the update stage, the same sample set is used to represent the state posterior in (32), i.e.,

$$X_k^{(p)} \leftarrow X_{k|k-1}^{(p)} \quad p = 1, \dots, P, \quad (35)$$

where \leftarrow denotes assignment.

The weights of these samples need to be adjusted using the measurement likelihood (as per the importance sampling

principle [28]), i.e.,

$$\zeta_k^{(p)} = \frac{\tilde{\zeta}_k^{(p)}}{\sum_{p'=1}^P \tilde{\zeta}_k^{(p')}}, \quad (36)$$

$$\tilde{\zeta}_k^{(p)} = \zeta_{k|k-1}^{(p)} l(\mathbf{Z}_k | X_k = X_k^{(p)}, \hat{\alpha}_k, \Delta \hat{\mathbf{t}}),$$

After finding the normalised weights in (36), we test degeneracy of the weighted particles. The degeneracy test is performed by finding the number of effective particles using

$$N_{eff} = \frac{1}{\sum_{p=1}^P \left(\zeta_k^{(p)} \right)^2}, \quad (37)$$

and, comparing it with a threshold \mathcal{T}_{eff} . When $N_{eff} < \mathcal{T}_{eff}$, we perform re-sampling (see, e.g., [27]) and continue filtering with a new, equally weighted sample set

$$\{\zeta_k^{(p)} \leftarrow 1/P, X_k^{(p)} \leftarrow \tilde{X}_k^{(p)}\}_{p=1}^P,$$

where $\{\tilde{X}_k^{(p)}\}$ is output by the re-sampler.

Using the above particle filter, the object state X_k at the k th CPI is estimated by using the empirical weighted average

$$\hat{X}_k = \sum_{p=1}^P \zeta_k^{(p)} X_{k|k-1}^{(p)}, \quad (38)$$

where \hat{X}_k denotes the estimated object state X_k .

A remarkable feature of the processing scheme driven by the Bayesian recursions above is that no fixed selection of the spatio-temporal steering vectors are used. The evaluation of the likelihood in the update stage in (36) specifies the steering vectors through (22) and (15) as a function of the state value $X_k^{(p)}$. Because $X_k^{(p)}$ are generated by sequential processing of the data cubes over CPIs, the resulting set of spatio-temporal steering vectors adapt to the measurements. This is in stark contrast with conventional processing chains in which the bearing and Doppler space is sampled with equal size steps leading to a fixed set of steering vectors and corresponding resolution bins. Thus, a super-resolution effect is achieved when finding the object locations as demonstrated in Section VI.

IV. MAXIMUM LIKELIHOOD ESTIMATION OF UNKNOWN PARAMETERS

In this section, we first introduce the ML estimator for the reflection coefficients. This estimator is an iterative algorithm realising Expectation Maximisation at each step of the recursive filtering detailed in Section III– in particular when evaluating the tracking update in (36)–and, is also central to long time integration detailed later in Section V. Second, we derive the ML synchronisation term estimator used together with the reflectivity estimator in the filter update in Section III.

A. ML estimation of the reflection coefficients

The reflection coefficients associated with an object at state X_k are unknown constants for the duration of a CPI and vary across consecutive CPIs due to the change of the object position, orientation etc. The likelihood of these reflectivities is

Algorithm 1 Particle EM algorithm for estimation of the reflection coefficients

- 1: Input: $\alpha^{(0)}, \epsilon \triangleright$ Initial guess and termination threshold
- 2: Input: $\{\zeta_{k|k-1}^{(p)} X_{k|k-1}^{(p)}\}_{p=1}^P \triangleright$ Particles from $p(X_k | \mathbf{Z}_{1:k-1})$
- 3: $i \leftarrow 1, \alpha^{(1)} \leftarrow \infty \triangleright$ Initialisation for the iterations
- 4: **while** $\|\alpha^{(i)} - \alpha^{(i-1)}\| > \epsilon$ **do** \triangleright Test convergence
- 5: Find $\hat{Q}(\alpha_k, \alpha_k^{(i-1)})$ in (44) using (42), (43) \triangleright **E step**
- 6: Find $\alpha^{(i)} \leftarrow \{\hat{\alpha}_{m,k}\}_{m=1}^M$ using (45), (43) \triangleright **M step**
- 7: $i \leftarrow i + 1$
- 8: **end while**
- 9: Return $\hat{\alpha}_k \leftarrow \alpha^{(i)}$

found by multiplying the likelihood in (29) with the priors for the other parameters and marginalising them out. Let us use the empirical prior for the synchronisation term (see, e.g., (31)) obtained using the ML estimator detailed in the next section. The likelihood to be maximised for estimation is hence found as

$$l(\mathbf{Z}_k | \alpha_k) = \int_X l(\mathbf{Z}_k | X_k, \alpha_k, \Delta \mathbf{t} = \Delta \hat{\mathbf{t}}) p(X_k | \mathbf{Z}_{1:k-1}) dX_k. \quad (39)$$

It is not straightforward to optimise this function due to the marginalisation involved. In ML problems involving such latent variables as the state variable X_k , the expectation maximisation (EM) method offers an iterative and gradient-free solution [16]. In this method, starting from an initial parameter configuration $\alpha_k^{(0)}$, an expectation that replaces the original likelihood is maximised. For the problem at hand, these iterations are given for $i = 1, 2, \dots$ by

$$\alpha_k^{(i)} = \arg \max_{\alpha} Q(\alpha_k, \alpha_k^{(i-1)})$$

$$Q(\alpha_k, \alpha_k^{(i-1)}) \triangleq E\{l(\mathbf{Z}_k | X_k, \alpha_k, \Delta \mathbf{t} = \Delta \hat{\mathbf{t}}) | \mathbf{Z}_{1:k-1}, \alpha_k^{(i-1)}\}$$

$$\propto \int_{X_k} \log l(\mathbf{Z}_k | X_k, \alpha_k, \Delta \hat{\mathbf{t}}) \times p(X_k | \mathbf{Z}_{1:k}, \alpha_k^{(i-1)}, \Delta \hat{\mathbf{t}}) dX_k. \quad (40)$$

Let us focus on the computation of the expectation in (40) and its maximisation. The state density function underlying the expectation is a state posterior conditioned on the previously found value of the reflectivities, i.e.,

$$p(X_k | \mathbf{Z}_{1:k}, \alpha_k^{(i-1)}, \Delta \hat{\mathbf{t}}) \propto l(\mathbf{Z}_k | X_k, \alpha_k^{(i-1)}, \Delta \hat{\mathbf{t}}) p(X_k | \mathbf{Z}_{1:k-1}), \quad (41)$$

where the density function on the right hand side is nothing but the predictive density of the Bayesian filtering recursions given in Section III. Thus, the samples generated in the prediction stage in (33) and (34) lead to an importance sampling [28] estimate of the expectation. Given $\{X_{k|k-1}^{(p)}, \zeta_{k|k-1}^{(p)}\}_{p=1}^P$, this importance sampling estimate is given by

$$\hat{Q}(\alpha_k, \alpha_k^{(i-1)}) \propto \sum_{p=1}^P \zeta_p^{(i-1)} \log l(\mathbf{Z}_k | X_k = X_{k|k-1}^{(p)}, \alpha_k, \Delta \hat{\mathbf{t}}), \quad (42)$$

$$\xi_p^{(i-1)} = \frac{l(\mathbf{Z}_k | X_k = X_{k|k-1}^{(p)}, \boldsymbol{\alpha}_k^{(i-1)}, \Delta \hat{\mathbf{t}}) \zeta_{k|k-1}^{(p)}}{\sum_{p'=1}^P l(\mathbf{Z}_k | X_k = X_{k|k-1}^{(p')}, \boldsymbol{\alpha}_k^{(i-1)}, \Delta \hat{\mathbf{t}}) \zeta_{k|k-1}^{(p')}} \quad (43)$$

where \hat{Q} denotes the estimate of the term proportional to Q in (40).

This approximation is a sum of terms quadratic in α_k . This can easily be seen by substituting from (29) and (22) to (42). The resulting expression is given in (44) (see the top of the next page). After taking the first order partial derivative of this expression with respect to $\alpha_{m,k}$ and setting it to zero, the ML estimate of the m th reflection channel is found in closed form given in (45) (see the top of this page).

Note that the ML estimator in (45) takes the inner product of the “whitened” measurements with the signal model \mathbf{s}_m given in (15) for each state particle $X_{k|k-1}^{(p)}$. This operation effectively performs digital beam-forming towards the particle state in space, and, matches its approach speed through its Doppler frequency encoded in \mathbf{s}_m . As a results, the estimator will not be interfered by other objects unless they appear very close to the state value in terms of the achievable spatial and Doppler resolution.

After finding $\hat{\boldsymbol{\alpha}}_k^{(i)} = \{\hat{\alpha}_{m,k}\}_{m=1}^M$ for M reflection coefficients using (45), convergence is tested by comparing the norm of the difference between the parameter configurations found in consecutive time steps with a threshold, i.e., iterations are terminated at i if

$$\|\boldsymbol{\alpha}_k^{(i)} - \boldsymbol{\alpha}_k^{(i-1)}\| < \epsilon,$$

where $\|\cdot\|$ denotes the complex Euclidean norm. A pseudo-code of these steps are given in Algorithm 1.

B. Synchronisation of the local processor with remote transmitters

In this section, we detail the ML estimation of the unknown synchronisation term Δt_m parametrising the time origin shift between the local receiver and the m th separately located transmitter. Our approach exploits the fact the data cube for the m th bi-static channel contains direct path signals from the transmitter that can be recovered by diverting a digital beam towards the transmitters spatial state simultaneously with other processing tasks on the data cube, e.g., those related to trajectory estimation and reflectivity estimation for other spatio-temporal points.

The direct path signal in the m th channel can easily be modelled using the spatial and temporal steering vectors defined in Section II in (7) and (13), respectively. The state of the m th transmitter is given by $X_m = [x_m, y_m, 0, 0]^T$ which is associated with the time-of-flight $\tau(X_m)$ given in (4) as the time to receiver. The angle of arrival is denoted by $\theta_m(X_m)$ which is computed using (6). Different from a reflection channel, the unknown reflectivity is replaced with a known pulse energy term. Thus, the CPI measurement vector at the r th range bin obtained by sampling the m th matched filter output is given by

$$\mathbf{Z}_m(r) = \sqrt{E_m} \tilde{\mathbf{s}}_m(r, X_m) + \mathbf{n}_m(r), \quad (46)$$

$$\tilde{\mathbf{s}}_m(r, X_m) \triangleq s(\Delta t_m) \mathbf{s}_s(\theta_m(X_m)) \otimes \mathbf{s}_t(\tau(X_m), \Omega_m(X_m) = 0)$$

$$\begin{aligned} & \times \Lambda_m(rT_p - \tau(X_m) - \Delta t_m), \\ & = s(\Delta t_m) s(\tau(X_m)) \times \mathbf{s}_s(\theta_m(X_m)) \otimes \mathbf{1} \\ & \times \Lambda_m(rT_p - \tau(X_m) - \Delta t_m), \end{aligned}$$

where E_m is the pulse energy, $\tilde{\mathbf{s}}_m$ is the noise free signal model associated with the transmitter state X_m , and, $\mathbf{1}$ is an $N \times 1$ all ones vector².

In the presence of reflectors, we will have received a superposition of this signal and reflections from different spatio-temporal states. In order to recover the direct path signal, a spatio-temporal steering vector that matches $\tilde{\mathbf{s}}_m$ in (46) is used which is given by

$$\begin{aligned} \mathbf{h}(X_m) & \triangleq \mathbf{s}_s(\theta_m(X_m)) \otimes \mathbf{s}_t(\tau(X_m), \Omega_m(X_m) = 0), \\ & = s(\tau(X_m)) \times \mathbf{s}_s(\theta_m(X_m)) \otimes \mathbf{1}. \end{aligned}$$

Note that this filter is nothing but a (scaled) beam-forming vector diverting a beam towards $\theta_m(X_m)$ and maps the $LN \times 1$ measurement vector $\mathbf{Z}_m(r)$ to a single complex value given by

$$\begin{aligned} d_m(r) & \triangleq \mathbf{h}(X_m)^H \mathbf{Z}_m(r) \\ & = LN \sqrt{E_m} s(\Delta t_m) \Lambda_m(rT_p - \tau(X_m) - \Delta t_m) \\ & \quad + n_m(r). \end{aligned} \quad (47)$$

Here, the noise term is the inner product of the beam-forming vector and the complex Gaussian measurement noise in (16), i.e.,

$$n_m(r) = \mathbf{h}(X_m)^H \mathbf{n}_m(r)$$

which itself is a random variable with a complex Gaussian distribution of mean zero and variance $\sigma_{d,m}^2 = \mathbf{h}(X_m)^H \Sigma_m \mathbf{h}(X_m)$.

As a result, the likelihood to be maximised is

$$l(d_m(1), \dots, d_m(\Gamma) | \Delta t) = \prod_{r=1}^{\Gamma} \mathcal{CN}(d_m(r); \mu_{d,m}(\Delta t), \sigma_{d,m}^2) \quad (48)$$

where the expected value of the complex Gaussian distributions as a function of Δt is given by

$$\begin{aligned} \mu_{d,m}(\Delta t) & = \sqrt{E_m} LN \exp(-j\omega_c \Delta t_m) \\ & \times \Lambda_m(rT_p - \tau(X_m) - \Delta t_m). \end{aligned} \quad (49)$$

Here, only those range bins for which the argument of Λ_m falls within $(0, 2T_p)$ contribute to the maximisation – otherwise, the corresponding distribution is same with that for the noise term. These range bins are given by

$$\tilde{\mathcal{E}}_m(\Delta t) = \begin{cases} \{r_m, r_m + 1\} & r_m T_p < \tau(X_m) + \Delta t_m \\ \{r_m\} & r_m T_p = \tau(X_m) + \Delta t_m, \\ \{r_m, r_m - 1\} & r_m T_p > \tau(X_m) + \Delta t_m \end{cases} \quad (50)$$

²Note that $\tilde{\mathbf{s}}_m$ differs from \mathbf{s}_m in (15) in that the latter uses the bi-static time-of-flight in both the temporal steering vector and the waveform auto-correlation delay, whereas, the former uses direct path time-of-flight. Because the transmitters are of zero Doppler frequency, the temporal steering vector reduces to an all ones vector scaled with $s(\tau(X_m))$.

$$\hat{Q}(\alpha_k, \alpha^{(i-1)}) = \sum_{p=1}^P \xi_p^{(i-1)} \left[\sum_{m=1}^M \sum_{r \in \mathcal{E}_m(X_{k|k-1}^{(p)})} \left(-\log(\pi^{LN} \det(\Sigma_m)) - \mathbf{Z}_{m,k}(r)^H \Sigma_m^{-1} \mathbf{Z}_{m,k}(r) \right. \right. \\ \left. \left. + 2 \operatorname{Re}\{\alpha_{m,k}^* \mathbf{s}_m(r, X_{k|k-1}^{(p)})^H \Sigma_m^{-1} \mathbf{Z}_{m,k}(r)\} - |\alpha_{m,k}|^2 \mathbf{s}_m(r, X_{k|k-1}^{(p)})^H \Sigma_m^{-1} \mathbf{s}_m(r, X_{k|k-1}^{(p)}) \right) \right] \quad (44)$$

$$\hat{\alpha}_{m,k} = \frac{\sum_{p=1}^P \sum_{r \in \mathcal{E}_m(X_{k|k-1}^{(p)})} \xi_p^{(i-1)} \mathbf{s}_m(r, X_{k|k-1}^{(p)})^H \Sigma_m^{-1} \mathbf{Z}_{m,k}(r)}{\sum_{p=1}^P \sum_{r \in \mathcal{E}_m(X_{k|k-1}^{(p)})} \xi_p^{(i-1)} \mathbf{s}_m(r, X_{k|k-1}^{(p)})^H \Sigma_m^{-1} \mathbf{s}_m(r, X_{k|k-1}^{(p)})} \quad (45)$$

where

$$r_m = \left\lfloor \frac{\tau(X_m) + \Delta t_m}{T_p} \right\rfloor.$$

Thus, the ML estimator that takes into account k data cubes at time k starting from the first one is given by

$$\Delta \hat{t}_m = \arg \max_{\Delta t_m} J_k(\Delta t_m) \\ J_k(\Delta t_m) = \log \prod_{k'=1}^k \prod_{r \in \tilde{\mathcal{E}}_m(\Delta t)} \mathcal{CN}(d_{m,k'}(r); \mu_{d,m}(\Delta t), \sigma_{d,m}^2) \\ \propto \sum_{k'=1}^k \sum_{r \in \tilde{\mathcal{E}}_m(\Delta t)} (d_{m,k'}(r) - \mu_{d,m}(\Delta t))^* \\ \times (d_{m,k'}(r) - \mu_{d,m}(\Delta t)). \quad (51)$$

Here, the relation between Δt and the objective function J_k is a concave relation on the average (and as k increases, asymptotically). However, (49) does not yield a closed form solution and render gradient-free iterative methods such as one-dimensional line search techniques [29] as better alternatives. These algorithms require only evaluation of (51) and iteratively reduce an initially selected interval of uncertainty.

We use the golden section search algorithm [29] and select and initial interval for Δt based on a preliminary search over the grid of values $\Delta t \in \{0, T_p, 2T_p, \dots, (\Gamma - 1)T_p\}$ which yields a rough estimate. Let us denote this term by $\Delta \hat{t}_0$ ³. The initial interval of uncertainty is selected as $[\Delta \hat{t}_0 - T_p, \Delta \hat{t}_0 + T_p]$. The golden section search reduces the width of this interval exponentially to a ratio of $(0.618)^{\nu-1}$ after ν iterations [29]. Therefore, in eight iterations, this width reduces below one tenth of a pulse duration, i.e., $T_p/10$. This search is detailed in Algorithm 2.

V. LONG TIME INTEGRATION FOR DETECTION

In this section, we detail the evaluation of the statistical test given in (17). The sufficient statistics of this test is given in Section II-C, in particular in (21)–(23). Here, we first combine the results from Sections III and IV into a single algorithm. Then, in Section V-A, we provide explicit formulae for finding the threshold as a function of a selected constant false alarm rate P_{fa} and integration time k .

In order to evaluate the likelihood ratio in (21), we first estimate $\Delta \mathbf{t}$ using Algorithm 2 for each bi-static channel.

³Note that, equivalently found is

$$\hat{r}_m = \left\lfloor \frac{\tau(X_m) + \Delta \hat{t}_0}{T_p} \right\rfloor.$$

Algorithm 2 Maximum likelihood estimation of Δt_m via golden section line search: The initial interval of uncertainty is selected as $[\Delta \hat{t}_0 - T_p, \Delta \hat{t}_0 + T_p]$ as detailed in Section IV-B.

```

1: Input:  $[\Delta t_1, \Delta t_2], \epsilon \triangleright$  Initial interval of uncertainty and
   termination threshold
2:  $\alpha \leftarrow 0.618$ 
3:  $\Delta \tilde{t}_1 \leftarrow \Delta t_1 + (1 - \alpha)(\Delta t_2 - \Delta t_1) \triangleright$  Evaluation point 1
4:  $\Delta \tilde{t}_2 \leftarrow \Delta t_1 + \alpha(\Delta t_2 - \Delta t_1) \triangleright$  Evaluation point 2
5: Compute  $J_k(\Delta \tilde{t}_1)$  and  $J_k(\Delta \tilde{t}_2)$  using (51), (49), (47)
6: while  $|\Delta t_2 - \Delta t_1| > \epsilon$  do  $\triangleright$  Until  $\epsilon$  accuracy is reached
7:   if  $J_k(\Delta \tilde{t}_1) > J_k(\Delta \tilde{t}_2)$  then
8:      $\Delta t_2 \leftarrow \Delta \tilde{t}_2 \triangleright$  New interval:  $[\Delta t_1, \Delta \tilde{t}_2]$ 
9:      $\Delta \tilde{t}_2 \leftarrow \Delta \tilde{t}_1, J_k(\Delta \tilde{t}_2) \leftarrow J_k(\Delta \tilde{t}_1) \triangleright$  Assignments
10:     $\Delta \tilde{t}_1 \leftarrow \Delta t_1 + (1 - \alpha)(\Delta t_2 - \Delta t_1)$ 
11:    Compute  $J_k(\Delta \tilde{t}_1)$  using (51), (49), (47)  $\triangleright$  New
    evaluation
12:   else
13:      $\Delta t_1 \leftarrow \Delta \tilde{t}_1 \triangleright$  New interval:  $[\Delta \tilde{t}_1, \Delta t_2]$ 
14:      $\Delta \tilde{t}_1 \leftarrow \Delta \tilde{t}_2, J_k(\Delta \tilde{t}_1) \leftarrow J_k(\Delta \tilde{t}_2) \triangleright$  Assignments
15:      $\Delta \tilde{t}_2 \leftarrow \Delta t_1 + \alpha(\Delta t_2 - \Delta t_1)$ 
16:     Compute  $J_k(\Delta \tilde{t}_2)$  using (51), (49), (47)  $\triangleright$  New
    evaluation
17:   end if
18: end while
19: if  $J_k(\Delta \tilde{t}_1) > J_k(\Delta \tilde{t}_2)$  then
20:   Return  $\Delta \hat{t}_m = \Delta t_1$ 
21: else
22:   Return  $\Delta \hat{t}_m = \Delta t_2$ 
23: end if

```

Given this quantity, we sequentially estimate the target trajectory \hat{X}_k (Section III) and the reflection coefficients $\hat{\alpha}_k$ using the EM iterations in Algorithm 1 over $k = 1, \dots, K$. As such, the integration of instantaneous likelihood ratios in (23) – given the aforementioned estimates – into the test value in (21) is carried out recursively. For this purpose, let us define the logarithm of the test value at k as

$$\log L_k \triangleq \sum_{k'=1}^k \sum_{m=1}^M \sum_{r \in \mathcal{E}_m(\hat{X}_k)} L(\mathbf{Z}_{m,k'}(r) | \hat{X}_{k'}, \hat{\alpha}_{m,k'}, \Delta \hat{t}_m) \\ = \log L_{k-1} + L(\mathbf{Z}_k(\hat{X}_k) | \hat{X}_k, \hat{\alpha}_k, \Delta \hat{t}) \quad (52)$$

where

$$L(\mathbf{Z}_k(\hat{X}_k) | \hat{X}_k, \hat{\alpha}_k, \Delta \hat{t}) \\ \triangleq \sum_{m=1}^M \sum_{r \in \mathcal{E}_m(\hat{X}_k)} L(\mathbf{Z}_{m,k'}(r) | \hat{X}_{k'}, \hat{\alpha}_{m,k'}, \Delta \hat{t}_m)$$

Algorithm 3 The proposed simultaneous tracking and long time integration algorithm

```

1: Input: Data cubes  $\mathbf{Z}_{m,k}$  for channels  $m = 1, \dots, M$ , time
   steps  $k = 1, \dots, K$   $\triangleright$  see (16)
2: Input: Detection threshold  $\mathcal{T}_K$ 
3: Initialisation: Generate particles in the cell under test
    $\{X_1^{(p)}, \zeta_1^{(p)}\}_{p=1}^P$   $\triangleright$  see (26)
4: Initialisation:  $\log L_0 \leftarrow 0$ 
5: for  $k = 1, \dots, K$  do
6:   if  $k \geq 2$  then  $\triangleright$  Prediction stage
7:     Generate  $\{X_{k|k-1}^p, \zeta_{k|k-1}^p\}_{p=1}^P$   $\triangleright$  see (33), (34)
8:   end if
9:   Find  $\hat{\Delta t}$  using Algorithm 2 for  $m = 2, \dots, M$   $\triangleright$  see
   Section IV-B
10:  Find  $\hat{\alpha}_k$  using the EM iterations in Algorithm 1
11:  Update  $\{X_k^{(p)}, \zeta_k^{(p)}\}_{p=1}^P$  using (35), (36)  $\triangleright$  Update
   stage
12:  Estimate  $\hat{X}_k$  using (38)
13:  Compute  $L(\mathbf{Z}_k(\hat{X}_k)|\hat{X}_k, \hat{\alpha}_k, \hat{\Delta t})$  using (53)
14:   $\log L_k = \log L_{k-1} + L(\mathbf{Z}_k(\hat{X}_k)|\hat{X}_k, \hat{\alpha}_k, \hat{\Delta t})$   $\triangleright$ 
   Integration step, see (52)
15: end for
16: if  $\log L_K > \log \mathcal{T}_K$  then  $\triangleright$  The detection test in (54)
17:   Return  $H_1$ 
18: else
19:   Return  $H_0$ 
20: end if

```

$$\begin{aligned}
&= \sum_{m=1}^M \sum_{r \in \mathcal{E}_m(\hat{X}_k)} \left(2\text{Re}\{\hat{\alpha}_{m,k}^* \mathbf{s}_m(r, \hat{X}_k)^H \Sigma_m^{-1} \mathbf{Z}_{m,k}(r)\} \right. \\
&\quad \left. - |\hat{\alpha}_{m,k}|^2 \mathbf{s}_m(r, \hat{X}_k)^H \Sigma_m^{-1} \mathbf{s}_m(r, \hat{X}_k) \right). \quad (53)
\end{aligned}$$

Here, (53) is the contribution of the measurements at time k into the integration in (52). The proposed processing performs coherent integration of $\mathcal{E}_m(\hat{X}_k) \times L \times N$ samples during a CPI in each channel. The integration is non-coherent across the channels as well as consecutive CPIs. The key is that the object trajectory is taken into account when performing all these simultaneously.

The object detection is hence performed by comparing the output of the aforementioned log-likelihood ratio to a detection threshold, i.e.,

$$\log L_K \underset{H_0}{\overset{H_1}{\gtrless}} \log \mathcal{T}_K, \quad (54)$$

where $\log \mathcal{T}_K$ is the detection threshold for a given constant false alarm rate (CFAR) for K steps of integration. The next section details the computation of this threshold value. A pseudo-code of the overall process is given in Algorithm 3.

A. Constant false alarm rate threshold for the detection test

In the hypothesis test in (54) it is highly desirable to select a threshold \mathcal{T}_K that yields a selected constant false alarm rate (CFAR) P_{fa} . For the calculation of \mathcal{T}_K as function of P_{fa} , we consider the distribution of the likelihood ratio given

in (21) under the $H = H_0$ hypothesis for the measurement in (16) [23]. Let us find the logarithm of the likelihood ratio after substituting from (23) into (21):

$$\eta_K \triangleq \log L_K = \sum_{k=1}^K \sum_{m=1}^M \sum_{r \in \mathcal{E}_m(X_k)} \eta_{m,k,r} \quad (55)$$

where the terms inside the summations are given by

$$\begin{aligned}
\eta_{m,k,r} &= 2\text{Re}\{\mathbf{s}_{m,k,r}^H \Sigma_m^{-1} \mathbf{Z}_{m,k}(r)\} - \mathbf{s}_{m,k,r}^H \Sigma_m^{-1} \mathbf{s}_{m,k,r} \\
\mathbf{s}_{m,k,r} &= \alpha_{m,k} \mathbf{s}_m(r, X_k). \quad (56)
\end{aligned}$$

The distribution of the real variable $\eta_{m,k,r}$ is a Gaussian when the signal model $\mathbf{s}_{m,k,r}$ is known and the measurements $\mathbf{Z}_{m,k}(r)$ are generated from a complex Gaussian [23, Chp.13], i.e., $\eta_{m,k,r} \sim \mathcal{N}(\cdot; \mu_{m,k,r}, \sigma_{m,k,r}^2)$, with the moments given by

$$\begin{aligned}
\mu_{m,k,r} &= -\mathbf{s}_{m,k,r}^H \Sigma_m^{-1} \mathbf{s}_{m,k,r}, \\
\sigma_{m,k,r}^2 &= 2\mathbf{s}_{m,k,r}^H \Sigma_m^{-1} \mathbf{s}_{m,k,r}.
\end{aligned}$$

Owing to the independence of the noise samples, η_K is also Gaussian for the case, i.e., $\eta_K \sim \mathcal{N}(\cdot; \mu_K, \sigma_K^2)$, with the moments given by

$$\mu_K = \sum_{k=1}^K \sum_{m=1}^M \sum_{r \in \mathcal{E}_m(X_k)} \mu_{m,k,r} \quad (57)$$

$$\sigma_K^2 = \sum_{k=1}^K \sum_{m=1}^M \sum_{r \in \mathcal{E}_m(X_k)} \sigma_{m,k,r}^2. \quad (58)$$

Therefore, the probability of false alarm P_{fa} is related to the test variable η_K in (55) and the threshold \mathcal{T}_K through

$$\begin{aligned}
P_{fa} &= \Pr\{\eta_K > \log \mathcal{T}_K | H = H_0\} \\
&= \int_{\log \mathcal{T}_K}^{+\infty} \mathcal{N}(\eta'_K; \mu_K, \sigma_K^2) d\eta'_K \\
&= Q\left(\frac{\log \mathcal{T}_K - \mu_K}{\sigma_K}\right)
\end{aligned}$$

where $Q(\cdot)$ denotes the tail probability function of the standard normal distribution [23]. As a result, the threshold \mathcal{T}_K given P_{fa} for K steps of integration is found as

$$\mathcal{T}_K = \exp(Q^{-1}(P_{fa})\sigma_K + \mu_K). \quad (59)$$

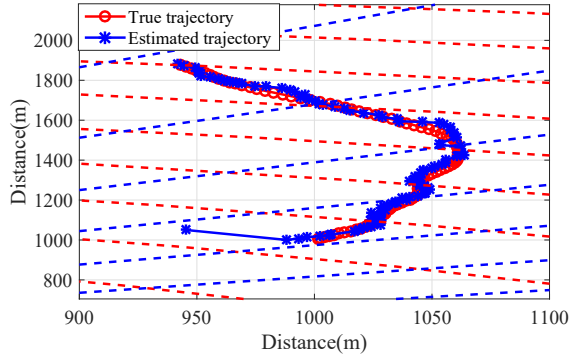
As a summary, a CFAR threshold for the proposed long time integration method is calculated using (56)–(59) given the true values of the reflectivities and the object trajectory specifying (56). This clairvoyant threshold is used in the next section, for comparing Algorithm 3 with the clairvoyant integrator and a conventional alternative.

VI. EXAMPLE

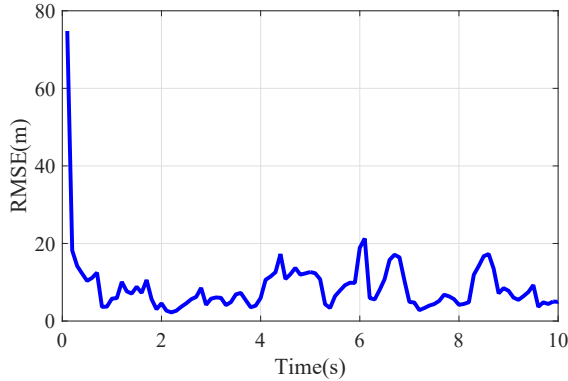
In this section, we demonstrate the proposed algorithm through an example and compare the efficacy of this approach with conventional techniques. We consider a scenario in which a ULA receiver co-located with a transmitter is at the origin of the 2D Cartesian plane, and, a separated transmitter is located at [10m, 500m] (see. Fig. 2). In this setting, $M = 2$ transmitters emit $N = 20$ linear frequency

TABLE I
TRANSMITTED SIGNAL PARAMETERS

Parameter	Value
Carrier frequency, i.e., f_c	10 GHz
Probing waveform bandwidth, i.e., B	1 MHz
Probing waveform duration, i.e., T_p	1.0 μ s
Pulse repetition interval (PRI), i.e., T	100 μ s
Number of range bins, i.e., Γ	100
Number of pulses, i.e., N	20
Number of elements in the ULA, i.e., L	20
Length of the coherent processing interval (CPI)	2 ms
Illumination period (Δ in (25))	0.1 s
Number of transmitters, i.e., M	2



(a) Typical scenario for an estimated trajectory



(b) Root mean square error (RMSE)

Fig. 4. Example scenario: (a) A low SNR (-6dB) object moves along the trajectory depicted with the red line. The estimated trajectory by the proposed algorithm is depicted with the blue line. The range bins resulting from sampling in time is shown by the dashed red lines. The bearing bins of the conventional processing chains is shown by the dashed blue lines. (b) RMSE of the estimated trajectory in (a).

modulated (i.e., up-chirp) waveforms towards a surveillance region and repeats this illumination pattern every 0.1 s. In this region, there is a low SNR object with an initial state $X_0 = [1000\text{m}, 1000\text{m}, 10\text{m/s}, 50\text{m/s}]$ moving along an unknown trajectory generated from the object dynamic model defined in (25). The ULA receiver with $L = 20$ elements collects measurements in accordance with the signal model in (16) from the local and the remote channels. Superpositioned in

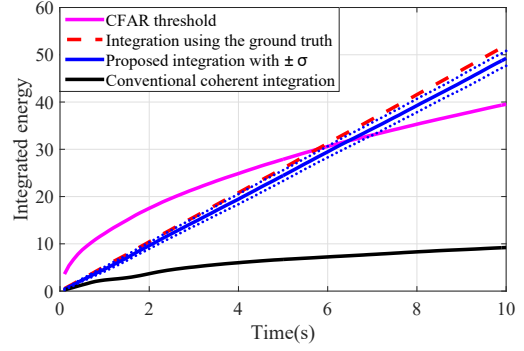


Fig. 5. Long-time integration using the proposed scheme, the clairvoyant integrator, and the conventional coherent integrator: The integrated sufficient statistics from the proposed integration averaged over 100 experiments is depicted by the solid blue line. The integrated value from the clairvoyant integrator is the dashed red line and the clairvoyant (CFAR) threshold (averaged for 100 experiments) is the solid magenta line. The conventional scheme leads to the solid black line.

the remote channel is the direct probing transmission from the transmitter. The parameter configuration of these transmissions are shown in Table I.

We simulate 100 independent sets of trajectories, reflection coefficients and array measurements. The reflection coefficients are generated from a complex Gaussian such that the expected SNR is -6dB in every channel. The direct signal from the non co-located transmitter is received with additive noise using (46) with an SNR of 0dB. The time reference shift of the remote transmitter and the receiver, i.e., Δt , is selected randomly in the range of $0 < \Delta t < PRI$ and used for all experiments.

We use Algorithm 3 for a time that spans 10 s and compare it with the following to detectors:

- 1) The clairvoyant detector: This detector uses the ground truth values of the unknown parameters (i.e., the object trajectories, reflectivities, and, the synchronisation term) when evaluating the likelihood ratio test in (21). The CFAR threshold for this detector is found using (56)–(59) as discussed in Section V-A.
- 2) Conventional coherent detector: This detector processes the measurements after mapping them over a grid of bearing and Doppler bins. These bins correspond to resolution cells which are found for the example system as follows: The bearing resolution is found as $\Delta\theta = 5.1^\circ$ using $\Delta\theta = \sin^{-1}\left(\frac{0.8192}{L}\right)$. The range resolution is found as $\Delta\tau = 150\text{m}$ using $\Delta\tau = \frac{c}{2B}$ (see, e.g., [22]). The velocity resolution of the conventional processing is found as $\Delta V = 7.5\text{m/s}$ by using $\Delta V = \frac{\lambda_c}{2NT}$ (or, equivalently, the Doppler resolution $\Delta\omega = 4\pi f_c \frac{\Delta V}{c} T$ as $\pi/10 \text{ rad s}^{-1}$). This detector integrates the mapped complex values for the same “cell under test” across time without taking account object manoeuvres [3].

We initiate our algorithm with $P = 400$ particles as a 20×20 uniform grid over a bounded region of location and velocity vectors such that the locations span the “cell under test”⁴. As the Bayesian filtering and trajectory estimation steps

⁴Note that because all the steering vectors during processing are selected by the Bayesian recursive filtering there is no fixed bearing or velocity resolution cells for our approach unlike the conventional detector.

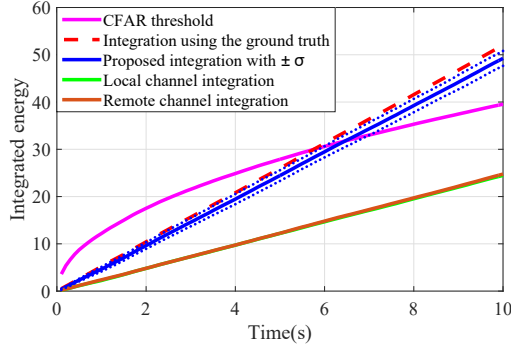


Fig. 6. The proposed scheme (solid blue line) versus the single channel integrations: The local channel (solid green line) integration, and the remote channel (solid brown line) integration fail to exceed the detection threshold.

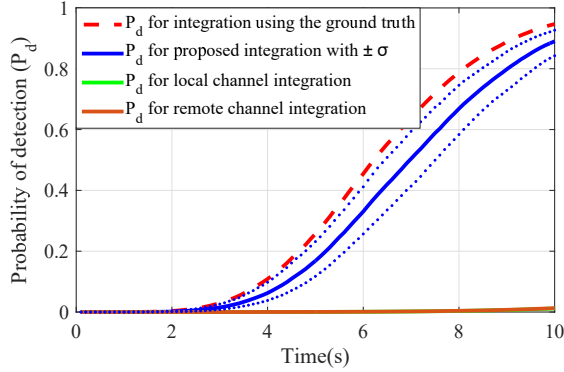


Fig. 7. Probability of detection (P_d) for the proposed scheme in comparison with the clairvoyant detector and the conventional detector.

iterate, these particles evolve to converge to the true state of the object simultaneously giving rise to the integrated value in (52). Fig. 4(a) illustrates a typical trajectory (red line) and the estimate output by the proposed algorithm (blue line) along with the resolution bins of the conventional detector Fig. 4(b) shows the root mean squared error (RMSE) of this estimate where the error reduces to a reasonably small value after only a few steps (with each step corresponding to a CPI). Note that the resolution bins of the system provides only a coarse view of the trajectory whereas the proposed algorithm yields a super-resolution effect.

Long time integration accuracy of the algorithm is coupled to the trajectory estimation performance through the EM iterations for finding the reflection coefficients (Algorithm 1). We compare the algorithm's integration with that of the clairvoyant and the conventional detector in Fig. 5. The clairvoyant integrator sets an upper bound for the integrated sufficient statistics, the average of which is depicted by the dashed red line. The proposed scheme performs very close to this bound (solid blue line rendering the average with $\pm\sigma$ bounds shown with dotted blue lines). Here, the proposed integration reaches to 49.24 at $t = 10$ s, which is relatively close to 51.78 achieved by the clairvoyant integration. The conventional scheme fails to continue the integration after the object leaves the initial cell under test. This integration is shown with the solid black line in Fig. 5.

The clairvoyant CFAR detection threshold is shown with the magenta line (averaged for the 100 experiments) in Fig. 5. The resulting detection using the integrated sufficient statistics

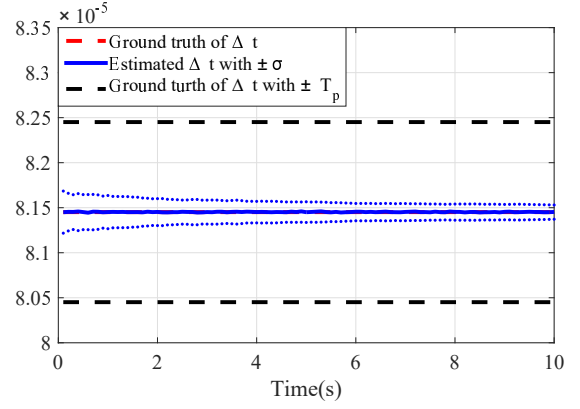


Fig. 8. Synchronised term estimation: Averaged synchronised term (solid blue line) estimated by using the proposed estimator versus the ground truth value (solid red line) with the bound the $\pm T_p$ bound (dashed black lines) of a preliminary search over the grids.

for each detector is performed by comparing its integration against this threshold. It is clearly recognised that the proposed scheme exceeds the CFAR threshold and enables us to decide on the object existence hypothesis ($H = H_1$) at $t = 6.5$ s whereas the conventional scheme stays in the region for the noise only signal hypothesis ($H = H_0$).

Fig. 6 compares the proposed integration with the integration using single channels. These values are found by using the proposed scheme with the local channel (solid green line) and the remote channel (solid brown line), respectively, and are averaged over 100 experiments. It is observed that all of the integrations (i.e., the proposed integration, the local channel integration, and the remote channel integration) increase over time, however, both the local channel and the remote channel integration fail to exceed the CFAR threshold by themselves.

Next, we consider the probability of detection P_d as a function of the integration time in Fig. 7. Here, the P_d of the proposed scheme (blue solid line) is found empirically and is averaged over the experiments. The P_d of the clairvoyant detector (dashed red line) sets the upper performance bound. The P_d of the proposed scheme increases with time and reaches $P_d = 0.89$ at $t = 10$ s, which is relatively close to the $P_d = 0.95$ of the clairvoyant detector. The P_d s of the local channel (solid green line) and the remote channel integration (solid brown line) stay in the vicinity of zero and fail to detect the object in an overwhelming majority of the experiments.

Let us consider the estimation of the time shift Δt in the remote channel. For this purpose, we use Algorithm 2 within Algorithm 3. Fig. 8 presents the averaged estimates (solid blue line) with $\pm\sigma$ bounds (dotted blue lines). We compare these values with the true value of Δt (red solid line). Also given are the $\pm T_p$ bounds (dashed black lines) for comparison. It is seen that the estimation error stays within a small fraction of the total pulse width T_p .

The benefits of the proposed scheme come with some additive cost of computations compared to conventional scheme. The computational cost of the proposed algorithm at the k th CPI is $P(N_X^2 + 2N_{I_1}N_{I_2}M(LN)^2)$ multiplications and $P(1 + 2N_{I_1}N_{I_2}M(LN - 1))$ additions, while the conventional detector requires M multiplications and $M(LN - 1)$ additions. Here, P is the number of particles, N_X denotes the dimen-

sionality of the object state in (25), and, N_{I_1} and N_{I_2} are the number of iterations for the EM algorithm in Algorithm 1 and the golden selection search in Algorithm 2, respectively.

VII. CONCLUSION

In this work, we have proposed a detection algorithm that performs the most efficient statistical test in order to detect manoeuvring and low SNR objects with an arbitrarily long time window of measurements. This test is carried out by simultaneous trajectory estimation and long time integration. Our approach can operate in mono-static, bi-static, and multi-static configurations, and, enables us to collect the entire evidence of object existence at the receiver by i) coherently integrating both mono-static and bi-static channels within a CPI, ii) performing non-coherent integration across different channels, and, iii) continuing integration for an arbitrarily long interval that contains many CPIs.

We have demonstrated that our approach can provide integration for an arbitrarily long interval with an effectiveness close to the best achievable by a clairvoyant integrator. As a result, this approach enables us to detect manoeuvring and very low SNR objects which cannot be detected by using conventional techniques.

APPENDIX A LIKELIHOOD LOCALITY

Let us consider the likelihood ratio test in (18) with the locality of the measurements $\mathbf{Z}_{m,k}(r) \in \mathcal{E}_m(X_k)$ to X_k . Let us define the complement of \mathcal{E} in the set of range bins, i.e., $\bar{\mathcal{E}} \triangleq \{1, 2, \dots, \Gamma\} \setminus \mathcal{E}_m(X_k)$. It can easily be seen that

$$l(\mathbf{Z}_{m,k}|X_k, \alpha_{m,k}, \Delta t_m, H = H_1) = \prod_{r \in \mathcal{E}_m} l(\mathbf{Z}_{m,k}(r)|X_k, \alpha_{m,k}, \Delta t_m, H = H_1) \prod_{r' \in \bar{\mathcal{E}}_m} p(\mathbf{Z}_{m,k}(r')). \quad (60)$$

Similarly, the likelihood for the noise-only signal hypothesis factorises as

$$l(\mathbf{Z}_{m,k}|X_k, H = H_0) = \prod_{r \in \mathcal{E}_m} l(\mathbf{Z}_{m,k}(r)|H = H_0) \prod_{r' \in \bar{\mathcal{E}}_m} p(\mathbf{Z}_{m,k}(r')), \quad (61)$$

which, after substituting into (18) with (60) leads to (21).

REFERENCES

- [1] M. Richards, W. Melvin, J. Scheer, J. Scheer, and W. Holm, *Principles of Modern Radar: Radar Applications*, ser. Electromagnetics and Radar. Institution of Engineering and Technology, 2014.
- [2] S. Haykin, "Cognitive radar: a way of the future," *IEEE Signal Processing Magazine*, vol. 23, no. 1, pp. 30–40, Jan 2006.
- [3] M. Richards, *Fundamentals of Radar Signal Processing*, ser. Professional Engineering. Mcgraw-hill, 2005.
- [4] X. Chen, J. Guan, N. Liu, and Y. He, "Maneuvering target detection via radon-fractional fourier transform-based long-time coherent integration," *Signal Processing, IEEE Transactions on*, vol. 62, no. 4, pp. 939–953, Feb 2014.
- [5] L. Kong, X. Li, G. Cui, W. Yi, and Y. Yang, "Coherent integration algorithm for a maneuvering target with high-order range migration," *IEEE Transactions on Signal Processing*, vol. 63, no. 17, pp. 4474–4486, Sept 2015.
- [6] Y. Boers and J. Driessen, "Multitarget particle filter track before detect application," *Radar, Sonar and Navigation, IEE Proceedings*, vol. 151, no. 6, pp. 351–357, Dec 2004.
- [7] E. Grossi, M. Lops, and L. Venturino, "A novel dynamic programming algorithm for track-before-detect in radar systems," *Signal Processing, IEEE Transactions on*, vol. 61, no. 10, pp. 2608–2619, May 2013.
- [8] H. L. Van Trees, *Detection, Estimation, and Modulation Theory: Radar-Sonar Signal Processing and Gaussian Signals in Noise*. Melbourne, FL, USA: Krieger Publishing Co., Inc., 1992.
- [9] S. Davey, M. Rutten, and B. Cheung, "Using phase to improve track-before-detect," *Aerospace and Electronic Systems, IEEE Transactions on*, vol. 48, no. 1, pp. 832–849, Jan 2012.
- [10] M. Uney, B. Mulgrew, and D. Clark, "Maximum likelihood signal parameter estimation via track before detect," pp. 1–5, Sept 2015.
- [11] O. Rabaste, C. Riche, and A. Lepoutre, "Long-time coherent integration for low SNR target via particle filter in track-before-detect," in *Information Fusion (FUSION), 2012 15th International Conference on*, July 2012, pp. 127–134.
- [12] K. Kim, M. Uney, and B. Mulgrew, "Detection of manoeuvring low SNR objects in receiver arrays," in *2016 Sensor Signal Processing for Defence (SSPD)*, Sept 2016, pp. 1–5.
- [13] —, "Simultaneous tracking and long time integration for detection in collaborative array radars," in *2017 IEEE Radar Conference (Radar-Conf)*, May 2017, pp. 0200–0205.
- [14] H. Godrich, A. Haimovich, and R. Blum, "Target localization accuracy gain in MIMO radar-based systems," *Information Theory, IEEE Transactions on*, vol. 56, no. 6, pp. 2783–2803, June 2010.
- [15] R. Niu, R. Blum, P. Varshney, and A. Drozd, "Target localization and tracking in noncoherent multiple-input multiple-output radar systems," *Aerospace and Electronic Systems, IEEE Transactions on*, vol. 48, no. 2, pp. 1466–1489, April 2012.
- [16] T. K. Moon, "The Expectation-Maximization Algorithm," *IEEE Signal Processing Magazine*, vol. 13, no. 6, pp. 47–60, Nov 1996.
- [17] B. Carlin and T. Louis, *Bayes and Empirical Bayes Methods for Data Analysis, Second Edition*, ser. Chapman & Hall/CRC Texts in Statistical Science. Taylor & Francis, 2010.
- [18] A. Haimovich, R. Blum, and L. Cimini, "MIMO radar with widely separated antennas," *Signal Processing Magazine, IEEE*, vol. 25, no. 1, pp. 116–129, 2008.
- [19] J. Li and P. Stoica, *MIMO Radar Signal Processing*. John Wiley & Sons, Inc., Hoboken, NJ, 2009.
- [20] A. De Maio and M. Lops, "Design principles of MIMO radar detectors," *Aerospace and Electronic Systems, IEEE Transactions on*, vol. 43, no. 3, pp. 886–898, July 2007.
- [21] Y. Yang and R. S. Blum, "MIMO radar waveform design based on mutual information and minimum mean-square error estimation," *IEEE Transactions on Aerospace and Electronic Systems*, vol. 43, no. 1, pp. 330–343, January 2007.
- [22] H. L. Van Trees, *Optimum Array Processing*. New York: Wiley-Interscience, 2002.
- [23] S. Kay, *Fundamentals of Statistical Signal Processing: Detection theory*, ser. Prentice Hall Signal Processing Series. Prentice-Hall PTR, 1998.
- [24] S. A. B. Ristic and N. Gordon, *Beyond the Kalman Filter: Particle Filters for Tracking Applications*. Artech House, 2004.
- [25] K. Murphy, *Machine Learning: A Probabilistic Perspective*, ser. Adaptive computation and machine learning. MIT Press.
- [26] M. Orton and W. Fitzgerald, "A Bayesian approach to tracking multiple targets using sensor arrays and particle filters," *Signal Processing, IEEE Transactions on*, vol. 50, no. 2, pp. 216–223, Feb 2002.
- [27] M. Arulampalam, S. Maskell, N. Gordon, and T. Clapp, "A tutorial on particle filters for online nonlinear/non-gaussian bayesian tracking," *Signal Processing, IEEE Transactions on*, vol. 50, no. 2, pp. 174–188, Feb 2002.
- [28] G. Casella and C. P. Robert, *Monte Carlo Statistical Methods*, 2nd ed. Springer, 2005.
- [29] H. D. S. M. S. Bazaraa and C. Shetty, *Nonlinear Programming*, 2nd ed. John Wiley & Sons, Inc., 1993.